

## Research Paper

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# Miniature drone antenna design for the detection of airliners

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## Abstract

In this paper, the design of a miniature antenna dedicated to the detection of airliners through the demodulation of Automatic Dependence Surveillance-Broadcast system (ADS-B) signals is presented. This antenna is designed for being embedded on the top of a drone in order to detect and avoid collisions with airliners. This antenna consists of an array of Planar Inverted-F Antennas, a quadrature feed network (FN) and a reflector plane (RP). The FN is designed to have output signals with the same amplitude and a 90° phase difference between each other. It achieves circular polarization and maintains the axial ratio of the antenna under –3 dB at the desired frequency (1.09 GHz). The antenna with the FN was manufactured and characterized. It weighs approximately 145 kg with its RP. The measured gain of the proposed antenna is about +3.7 dBi. To validate the design, the manufactured antenna was tested with a Universal Software Radio Peripheral for the processing of ADS-B signals at the French National Microwaves Days 2019 (JNM) student contest. The detection of airliners can reach up to 437 km.

## Introduction

With the emergence of drones in the last years, new problems arise especially concerning the sharing and the safety of the airspace. Thus, the detection and the localization of aircraft by unmanned aerial vehicles are essential to avoid collisions. The Automatic Dependence Surveillance-Broadcast (ADS-B) surveillance technology is more and more used by aircraft to broadcast periodically some information like their position to the air traffic control stations and/or to other aircraft. It is above all a cheap, simple, and reliable alternative to using expensive Radar technologies to locate aircraft from the ground. The position of aircraft is no more calculated and approximated from the Radars data but sent by the aircraft themselves from embedded Global Positioning System. Nevertheless, the transmitted unscrambled information like location can be easily received and used by non-targeted users like drones or radio-amateurs.

Within the framework of the 21<sup>st</sup> French National Microwaves Days (JNM) at Caen (France) in 2019, a student contest was organized to highlight the best miniature receiving antenna for drones dedicated to the detection, localization, and avoidance of airliners by the use of ADS-B signals. During this contest, different antennas were tested and the most efficient was defined as the one, which has detected the farthest aircraft.

The signal processing needed to parse ADS-B messages and locate the aircraft during this competition was processed thanks to the Software Defined Radio (SDR) solution presented in [1, 2].

This paper presents the winning antenna, which is a compact array of four three-dimensional Planar Inverted-F Antenna (PIFA). According to the constraints imposed by the contest organization, this antenna is fully passive, has a maximal size inferior or equal to 80 mm × 80 mm × 20 mm, placed above the center of a 200 mm × 150 mm reflector plan emulating the top of a drone and connectable through a female SMA connector. This miniature antenna is tuned to be effective in the ADS-B band used in Europe, which is around 1090 MHz. It also has nearly circular polarization, a high aperture angle of 97° and a gain of +3.7 dBi.

## Proposed antenna design

For the detection of ADS-B signals, patch and quasi-omnidirectional antennas are generally used. In [1], a circularly polarized patch antenna by corner cut dedicated to ADS-B systems is presented. In [3], this is a compact T-Slot patch antenna, which is presented for the same application. In order to detect the farthest aircraft, a quasi-omnidirectional and circularly

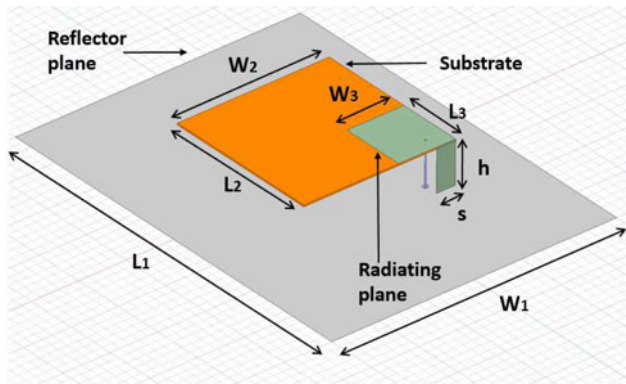
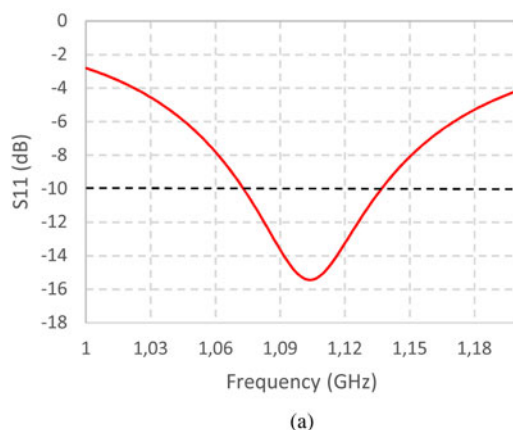


Fig. 1. Configuration of the antenna with single PIFA.

Table 1. Geometrical parameters of a single PIFA antenna

Parameter	Value (mm)
Length of the ground plane ( $L_1$ )	200
Length of the substrate ( $L_2$ )	80
Length of the radiating plane ( $L_3$ )	$\lambda/4 \mid \sim 68.75$
Width of the ground plane ( $W_1$ )	150
Width of the substrate ( $W_2$ )	80
Width of the radiating plane ( $W_3$ )	30
Height ( $h$ )	20
Shorting pin width ( $s$ )	10

polarized antenna seemed to us the appropriate choice. Moreover, the designed antenna should have good efficiency and a gain as high as possible. Thus, the proposed design consists of an array of four PIFA. Thanks to the compactness of each PIFA, it is possible to place four of them in the specified area and, thus, increase the global gain of the array. For instance, the design of a PIFA array with circular polarization for Nanosatellite application is proposed in [4]. Its size is about 100 mm  $\times$  100 mm and has a gain of +4.78 dBi at 2.35 GHz.



## Antenna design

First, a single PIFA was simulated and optimized with HFSS software. Its geometry in Fig. 1 consists of a radiating element short-circuited to a ground/reflector plane (RP) by a shorting pin and a feeding point placed between the two planes (and close to the shorting pin). The RP size and position were chosen according to the contest size requirements. It has a dimension of 150 mm  $\times$  200 mm and the antenna size (without reflector) should not exceed 80 mm  $\times$  80 mm with a maximum height of 20 mm above the reflector.

The antenna was designed on a FR4 substrate (thickness: 0.8 mm, dielectric constant: 4.4, dielectric loss tangent: 0.02). Table 1 shows the dimensions of the designed single PIFA that is not centered on the ground plane, thus anticipating the mounting position in the antenna array.

The position of the feed point is controlled to have an input impedance of 50  $\Omega$  at the targeted frequency. The measured reflection coefficient as a function of the frequency plotted in Fig. 2(a) shows a good matching with less than -15 dB at 1.09 GHz. As seen in the radiation pattern in Fig. 2(b), the maximum gain of +4 dBi at 1.09 GHz was obtained for an elevation angle of 60° or equivalently for  $\theta = 30^\circ$ ; where theta is the angle relative to the Oz-axis that is perpendicular to the antenna plane (i.e. the horizontal or xOy plane).

An array of four PIFA was designed by adding three other identical single PIFA at each corner of the FR4 substrate as presented in Fig. 3. Circular polarization was obtained with the relative phases of 0°, 90°, 180°, and 270° at the source ports.

With previous parameters depicted in Table 1, the antenna array does not operate at the targeted frequency. Moving the feeding point at the center of each single patch surface (radiating plane), allows reaching a good matching at 1.09 GHz. The radiating surface has also been modified to a square ( $W_3 = L_3 = \lambda/4$ ).

Figure 4(a) shows the reflection coefficient of the array of PIFA at each input port. The port 1 and 3 with a respective excitation of 0° and 180° give the same simulated value of the reflection coefficient while the ones at the port 2 (90°) and port 4 (270°) are the same.

As depicted in Fig. 4(b), the miniature antenna array with quadrature feed can reach a maximum gain of +4.85 dBi oriented over the Oz-axis that is perpendicular to the antenna plane.

The rotation of the antenna array above the RP has an important impact on the axial ratio (AR). As a result of an optimization

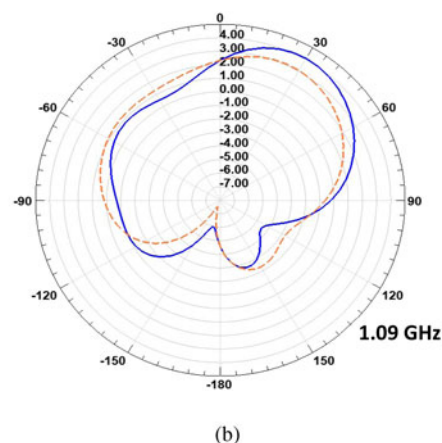


Fig. 2. (a) Simulated reflection coefficient and (b) radiated pattern of a single PIFA antenna ( $xOz/\varphi = 0^\circ$  cut – blue solid line – and  $yOz/\varphi = 90^\circ$  – orange dash line –).

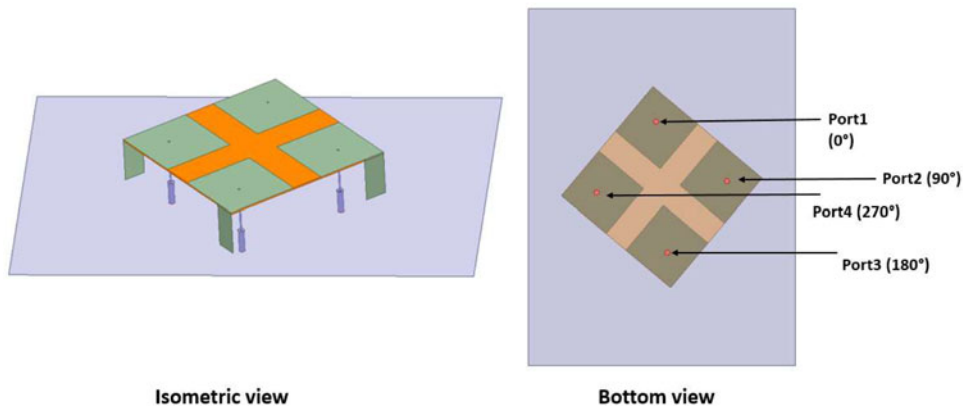


Fig. 3. Isometric and bottom view of the array PIFA antenna.

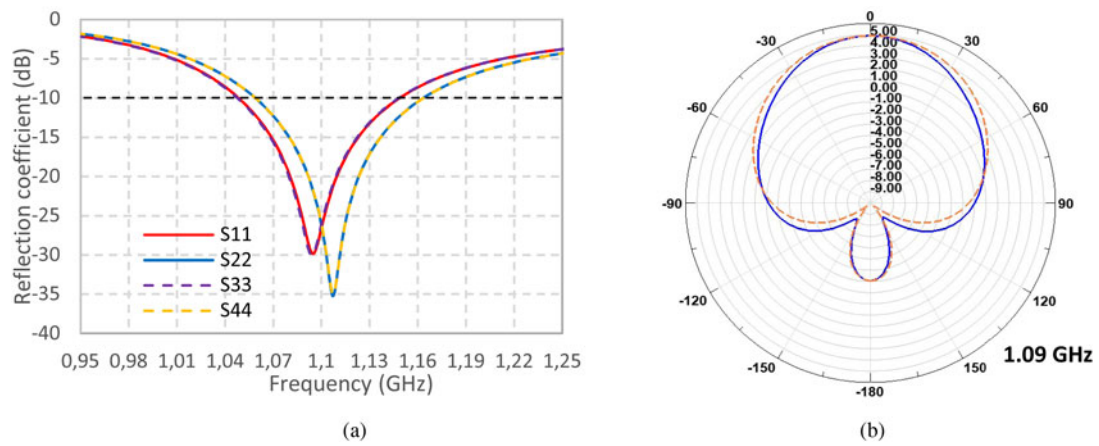


Fig. 4. (a) Simulated reflection coefficient at each port of the antenna; (b) the radiation pattern of the PIFA antenna array in two vertical planes:  $xOz/\varphi=0^\circ$  (blue solid line) and  $yOz/\varphi=90^\circ$  (orange dash line).

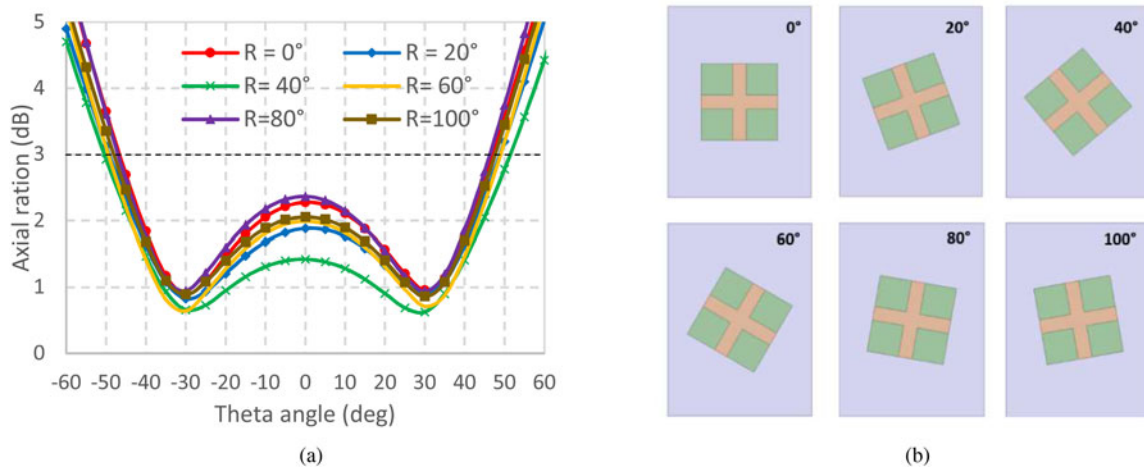


Fig. 5. (a) Variation of the axial ratio as function of the antenna rotation angle at  $\varphi=0^\circ$  and 1.09 GHz; (b) antenna rotation angle with the respect to the reflector.

process, we found that the rotation of the array about  $40^\circ$  on the RP improves the most the AR. As seen in Fig. 5(a), the antenna has an AR bandwidth ( $AR < 3$  dB) about  $[-50^\circ; 50^\circ]$ .

### Feed network

Instead of using four ports to feed the four PIFA, we decide to design a feed network (FN) based on the circuitual model

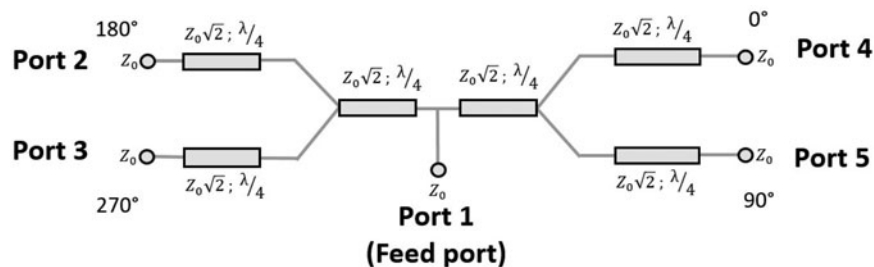


Fig. 6. Circuit model of the feed network.

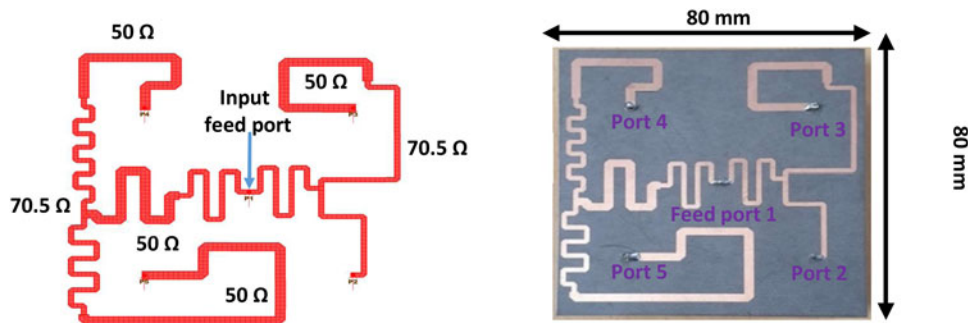


Fig. 7. Layout and photograph of the fabricated quadrature feed network.

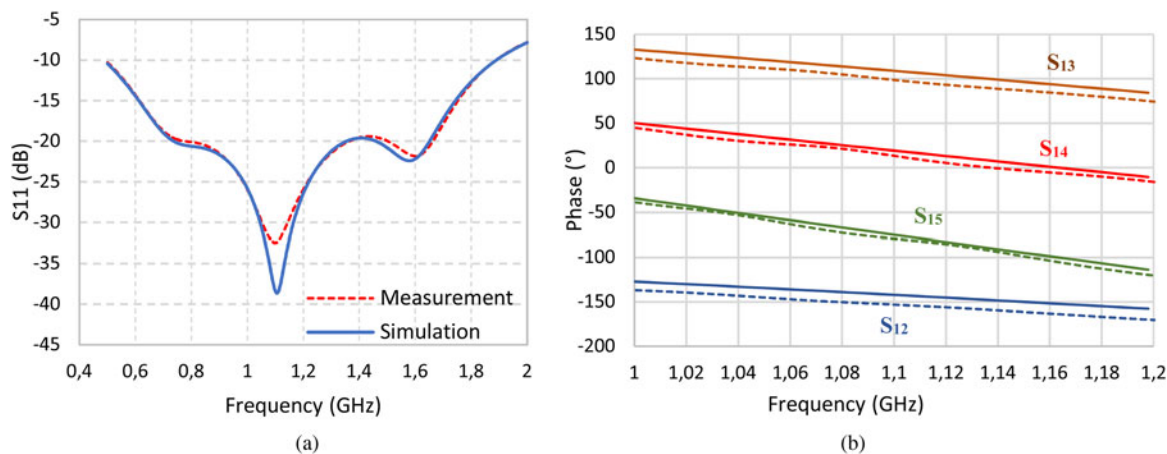


Fig. 8. (a) Simulated (line) and measured (dashed line) of the reflection coefficient (at the input of the feed port); (b) The phase delay between port 1 and each of the others.

represented in Fig. 6. It was designed to have equal 6 dB power splitting and create 90° phase shift between each output signal.

Different structures of the feeding network are proposed in the literature. The system designed in [5] is a combination of Wilkinson power dividers and two branch-line couplers in the 2.5 GHz band. Another design is proposed in [6], consists of three T-junction in phase power dividers for 30 GHz frequency band. It exists a sequential rotation network (SRN) too, which could answer to our constraints [7]. Figure 7 shows the chosen layout of the feeding network. It consists of a combination of transmission lines with different characteristic impedances and lengths connected by using a T-junction. They are used as an impedance transformer and a 90° phase shifter. Additionally, microstrip meander lines were used for the compactness in

order to respect the specified area. Advanced Design System (ADS) Keysight software was used for the design, the simulation, and the optimization of this circuit. For accurate simulation, a Momentum simulation was performed. As presented in Fig. 7, the length of 50 Ω line was adjusted to  $\lambda/4$  to allow the targeted phase shift between each antenna port.  $Z_0$  corresponds to the equivalent impedance of the connected antenna in the port (50 Ω).

The FN was fabricated on Duroïd substrate (substrate thickness: 0.787 mm, relative permittivity: 2.3 and loss tangent: 0.0012) to allow lower losses. For the measurement, the four end-line ports were connected to a 50 Ω load to replace the impedance of the miniature antenna.

The input reflection coefficient of the circuit is presented in Fig. 8. The agreement between measurement and simulation is

**Table 2.** Measured power split and phase at each port of the feed network at 1.09 GHz

S-Parameters	$S_{12}$	$S_{13}$	$S_{14}$	$S_{15}$
Measured power split (dB)	-6.1	-6.94	-5.93	-5.69
Simulated phase (degree)	-140.3	111.9	22.8	-70.4
Measured phase (degree)	-151.8	101.7	17.5	-76.6

**Table 3.** Measured phase difference between output ports of the feed network at 1.09 GHz

Output Port	$S_{14}-S_{15}$	$S_{14}-S_{12}$	$S_{14}-S_{13}$
Phase difference (degree)	94.1	169.3	-84.2 (modulo $2\pi$ ) = 275.8

very good and the input reflection coefficient is less than -15 dB for the targeted frequency.

The measured power division/split shown in Table 2 is approximately equal to 6 dB at each output port. The slight difference is mainly due to the compact size of the feeding network that does not allow a fully optimized design in the targeted area that is mainly identical with the maximum allowed surface

(according to the JNM contest) for our antenna array (80 mm × 80 mm × 20 mm).

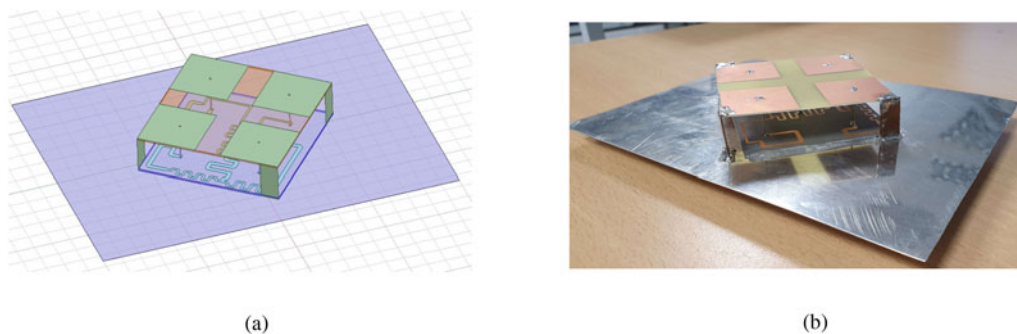
The port 4 with the lowest phase was chosen as the port reference of the FN. As depicted in Table 3, the FN satisfies a progressive 90° phase delay for other ports. The maximum phase error obtained is about +16.5° for the port 2. This difference is due to the last line connected to port 2, which is not exactly equal to  $\lambda/4$ .

**Experimental results and discussion**

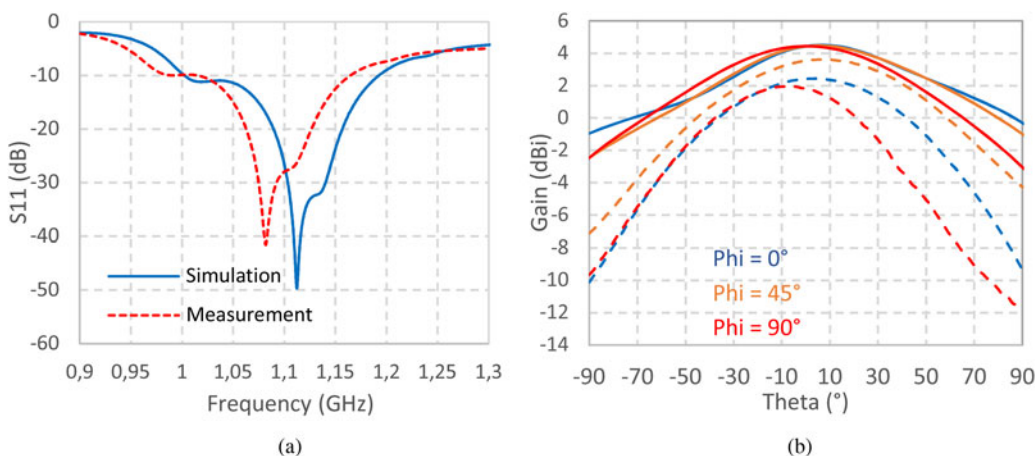
A prototype of the array of PIFA with the F was manufactured and characterized (Fig. 9). The FN is placed at the top of the RP to have a compact prototype. In order to place the SMA connector and to avoid the short-circuit of the feeding point and the ground plane, a small gap was added between the RP and the circuit. With its RP and FN, the antenna array weighs approximately 145 g.

As seen in Fig. 10(a), a good correlation of the  $S_{11}$  between simulation and measurement was obtained with a frequency shift of only 30 MHz.

The measured gain presented in Fig. 10(b) was performed in an anechoic chamber. The  $\varphi$  angle corresponds to the cut plane of the RP.  $\Phi$  angle equal to 45° means that the PIFA array without RP is perpendicular to the  $xOz$  plane. The measured result shows that the antenna has a maximum gain of +2.44, +3.7, and +1.94 dBi respectively, at  $\varphi = 0^\circ$ ,  $\varphi = 45^\circ$ , and  $\varphi = 90^\circ$ .



**Fig. 9.** (a) Designed antenna on HFSS Software; (b) Photograph of the manufactured antenna.



**Fig. 10.** (a) Measured (dashed line) and simulated (continuous line) reflection coefficient at the input of the feed port of the PIFA array; (b) the simulated (continuous line) and measured (dashed line) gain for three vertical cut planes ( $\varphi = 0^\circ$ ,  $\varphi = 45^\circ$ , and  $\varphi = 90^\circ$ ).



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