### Target technology development for the research of high energy density physics and inertial fusion at the RFNC-VNIIEF

V.M. IZGORODIN, F.M. ABZAEV, A.P. BALYAEV, A.V. BESSARAB, I.N. CHERKESOVA, V.V. CHULKOV, D.YU. FENOSHIN, S.G. GARANIN, V.G. GOGOLEV, A.G. GOLUBINSKY, YU.V. IGNAT'EV, D.A. IRINICHEV, A.E. LACHTIKOV, A.P. MOROVOV, V.V. NAZAROV, G.P. NIKOLAEV, A.P. PEPELYAEV, A.V. PINEGIN, I.M. ROJZ, V.N. ROMAEV, E.YU. SOLOMATINA, M.G. VASIN, AND A.V. VESELOV

The Russian Federal Nuclear Centre-All-Russia Research Institute of Experimental Physics (RFNC-VNIIEF), Nizhniy Novgorod region, Russia

(RECEIVED 8 August 2009; ACCEPTED 19 August 2009)

#### Abstract

Results in some directions of the target technology for research on high energy density and laser fusion at the Russian Federal Nuclear Centre–All-Russia Research Institute of Experimental Physics for the last three years are presented. The results of development of optical and X-ray methods of characterization and manufacturing techniques of targets for studying the equation-of-state at high pressures and the condensed rare gas targets for the influence of pulse-repeated laser irradiation are given.

**Keywords:** High density plasma physics; Inertial fusion; Jet generator for nanolithography; Laser fusion target; X-ray fluorescence and absorption analysis

#### 1. INTRODUCTION (S. G. Garanin)

This review will concentrate entirely on the target research and development at the Russian Federal Nuclear Centre– All-Russia Research Institute of Experimental Physics (RFNC–VNIIEF). The recent progress in target technology at other laboratories engaged in fusion research with laser and particle beams is documented in an impressive amount of detail (Aleksandrova *et al.*, 2008; Borisenko *et al.*, 2008; Chatain *et al.*, 2008; Cook *et al.*, 2008; Fernandez *et al.*, 2005; Koresheva *et al.*, 2009; Meyertervehn *et al.*, 1990; Tahir *et al.*, 2008; Yang *et al.*, 2008).

Research on the physics of high-temperature laser plasma with respect to laser fusion has actively been carried out at the RFNC–VNIIEF since 1970. Experiments in the following basic directions (Il'kaev & Garanin, 2006) are being developed at high-power monopulse laser installations ISKRA-3 (Gaidash *et al.*, 1974), ISKRA-4 (Cormer, 1980), ISKRA-5 (Annenkov *et al.*, 1991), BEAM (Besnasiuk *et al.*, 2002), FEMTO (Andreev *et al.*, 2004): (1) dynamics of compression of direct and indirect irradiation fusion targets; (2) influence of target irradiation asymmetry on output parameters (three-dimensional (3D) compression of fusion fuel, temperature of compressed deuterium-tritium (DT) gas, and neutron output); (3) X-radiation transfer into the indirect irradiation targets; (4) the equation-of-state of substances; (5) influence of the turbulent mixing; and (6) physics of X-ray laser operation.

One of the key problems in the realization of these research directions is the problem of precision specialized targets of various types. The characteristic area sizes of focused laser radiation, the conditions needed for research in the physics of high density energy are created, are in the range from tens of microns up to several millimeters. Thus, for reception of the detailed information on dynamics of shells acceleration, evolution of gas dynamic instabilities, physics of generation, and translation of X-radiation, the accuracy of manufacturing of laser target components at a level of about 10 nm and a roughness of 1 to 5 nm are necessary. Moreover, rigid demands are needed for density

Address correspondence and reprint requests to: V.M. Izgorodin, The Russian Federal Nuclear Centre–All-Russia Research Institute of Experimental Physics (RFNC-VNIIEF), 607190 Sarov, Mira Street 37, Nizhniy Novgorod region, Russia. E-mail: izgorodin@otd13.vniief.ru

homogeneity of materials, which targets are produced from. Creation of compound targets, inclusion of their structure of elements with controllable asymmetry of size and heterogeneity of layers, represents special complexity.

The above explains the importance of the development of technology of targets manufacturing for laser experiments and creation of precision quality monitoring of their parameters. It is necessary to also note the increasing rate of fire of laser installations and the need in quantity of produced targets.

Research at the RFNC-VNIIEF is based on the development of modern technologies for the creation of various types of targets (Veselov *et al.*, 1995; Andramanova *et al.*, 1999). Manufacturing techniques of spherical shells, filled with DT, manufacturing of indirect irradiation spherical, and cylindrical targets, various multi-layered film targets etc. have been developed. Taking into account the development of a new generation installation, ISKRA-6 (Galakhov *et al.*, 1999), the creation of "targets factory," which will provide researchers with demanded targets for experiments is planned on the basis of existing technologies. A number of technologies developed at the RFNC-VNIIEF are presented below.

In Section 2, nondestructive control of transparent shell thickness is given. This method is based on a simultaneous measurement of external and internal surface positions, by means of scanning white light interferometer. In Sections 2.1 and 2.1, results of spectral X-ray application techniques for the control of target parameters are presented. In Section 3, the substantiation use of fluorescence and absorption methods for measurement of thickness of multi-layered and multicomponent films, and the analysis of a mistake, and the sensitivity of measurements are given. Results of fluorescence analysis for definition of gas structure in targets and T concentration are summarized. In Section 4, the illustration of application of measurements of the X-ray characteristic and braking radiation caused by β-radiation of T for the definition of its surface concentration in materials is given. This method can be used for studying processes of hydrogen saturation and desorption in construction materials.

Section 5 is devoted to manufacturing techniques of targets. In the first of the procedures of target manufacturing for the research of substance equation-of-state at high pressures by influence of a powerful laser impulse on a sample surface is described. One of the main requirements to such targets is a high accuracy of layers thickness measurement and small surfaces roughness. Results of work in these directions are given for manufacturing targets from Al, Cu, Pb, and polymeric films. In Section 6, results of activities on creation of a target from the condensed rare gases (Ar, Kr, and Xe) are described. The target was prepared in the form of a continuous liquid jet and can be used for the influence of the pulse-periodic laser with frequency up to hundred Hertz. Exited X-ray radiation can be used in microelectronics for illumination of nanolithography.

#### 2. NONDESTRUCTIVE CONTACTLESS CHARACTERIZATION OF AN INTERNAL SURFACE OF TRANSPARENT SHELLS (A. V. Veselov, A. G. Golubinsky, E. Y. Solomatina)

For inertial confinement fusion, it is important to know power spectra of inner and outer radii of shell surfaces. Usually, atomic force microscopy is used for measurements of outer surface (Cook, 1994). However, this method is not acceptable for nondestructive measurements of the inner surface. In this section, we present a method of simultaneous measurements of inner and outer surfaces of transparent shells by profiler based on scanning white light interferometry.

#### 2.1. Surface Measuring Instrument

3D scanning interferometry of white light is widely described (Deck & de Groot, 1994; Freischlad & Koliopoulos, 1990; de Groot *et al.*, 2002). In our research, we used a device based on the Linnik interferometer for shell measuring. Construction of such a device is shown in Figure 1. The use of this construction allows partially compensate dispersion of light in the shell material in the case of large thickness. Measuring is carried out by vertical shell displacement. Interference figures are recorded by camera where the interferometer arms are approximately equal. Light intensity produces a signal in every pixel of the figure, like the signal in Figure 2. From the peak location of the signal envelope, it is possible to reconstruct the outer and inner surface of the transparent shell. The typical shape of the shell is shown in Figure 3.

#### 2.2. Results

Shells of different diameters (*D*) and thicknesses ( $\Delta$ ) were used for measurements (Table 1). Nonuniformity of thickness was less then 5%. After reconstruction of outer and inner shell surfaces radii (*R*), centers and roughness ( $\sigma$ ) of surfaces were calculated. The difference between sphere centers of inner and outer surfaces was less then 0.2 µm. It was less then the nonuniformity thickness of the shells. Thus, we have shown the possibility of the characterization of inner surface of the transparent shell. Further device improvements will be needed to increase the range of the measured parameters.

#### MEASUREMENT OF LASER FUSION TARGETS PARAMETERS BY X-RAY SPECTRAL METHODS (M. G. Vasin, A. V. Veselov, Yu. V. Ignat'ev, V. M. Izgorodin, A. E. Lachtikov, A. P. Morovov, V. V. Nazarov, A. V. Pinegin, V. N. Romaev)

X-ray spectral techniques possess a number of advantages and are applied in various measurements of element composition, optical thickness of substance, geometrical



Fig. 1. Schematic diagram of the profiler based on scanning white light technique and the Linnik interferometer MII-4.

parameters, etc. Practically, all methods of the analysis can be applied to the problem of characterization of spherical, cylindrical, and flat targets used in experiments with laser plasma. For universal certification, it is optimal to use the techniques of the analysis developed on the basis of fundamental parameters (Mantler, 1987). X-ray spectral techniques allows us to define simultaneously element composition and thickness of a film (Laguiton & Parrich, 1977; Nazarov, 1991), to develop the analysis of multilayered samples in view of inside- and inter-layer matrix effects (Mantler, 1986; Huang & Parrich, 1986), to measure composition and component quantity of a gas mix in spherical targets (Weinstein & Weir, 1980; Ignat'ev *et al.*, 2002), and composition of the shells. At its realization, it is not required to have plenty of standard samples.



Fig. 2. Intensity from single pixel during scan.

### 3.1. Measurement of Gas Amount in Spherical Targets

The measurement scheme of gas amounts in microspheres is presented in Figure 4. At measurement with external excitation of fluorescence, the microsphere is placed in a twochannel vacuum chamber for exclusion of parasitic fluorescence of air. Exciting radiation of the X-ray device can serially pass through both of the channels. One of the channels is intended for background measurements, and the second channel is the worker, the investigated microsphere is placed on its axis in the chamber. For removal of fluorescent radiation from the detector, the hole in the center of a face wall of the chamber is prepared. All holes were closed by a thin lavsan film. If the target contains T, the X-ray device is not required as fluorescent radiation is excited by the  $\beta$ -electrons of T.

#### 3.1.1. Targets Filled with T

Spectra of X-ray radiation of glass microsphere containing T and Ar are presented in Figure 5. Peaks of Si, K, Ar, and



Fig. 3. Reconstructed shell surfaces ( $100 \ \mu m \times 70 \ \mu m$ ). (a) The interference pattern on external and internal surfaces of the shell of 330  $\ \mu m$  diameter and 3.1  $\ \mu m$  thick wall. (b) 3D image of a segment of the shell with 330  $\ \mu m$  diameter and 3.1  $\ \mu m$  thick wall. (c) 3D image of a segment of the shell with 310  $\ \mu m$  diameter and 3.6  $\ \mu m$  thick wall.

Table	1.	Parameters	of shells
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N	D (µm)	$\Delta$ (µm)	$R_{\rm in}$ (µm)	$\sigma_{in}\left(\mu m\right)$	$R_{\rm out}$ (µm)	$\sigma_{out}$ (µm)
1	230	4.9	113	0.005	118	0.03
2	240	7.7	115	0.003	123	0.01
3	280	11.4	133.5	0.002	144	0.01

a braking spectrum are visible in the figure. Si and K are components of the glass shells.

The quantity of T is based on the knowledge of glass composition, geometrical measurement, and literary data about an electron energy distribution, section of excitation of X-ray radiation, and its absorption in substance. As a base expression, the expression for the quantum amount of the analytical falling-line of a microsphere wall on the detector in time units are shown below,

$$\dot{N} = \frac{1}{4\pi} \int_{0}^{2\pi} d\Phi \int_{0}^{X} x \cdot dx \int_{0}^{2\pi} d\varphi \int_{0}^{\pi} \sin \vartheta \cdot d\vartheta \int_{r_{0}}^{r_{0}+d} w \cdot \frac{L - r \sin \vartheta \sin \varphi}{l^{3}}$$
$$\times \exp\left(-\sum \mu_{i} l_{i}\right) \cdot r^{2} dr, \qquad (1)$$



Vacuum chamber

Fig. 4. The scheme of measurements of gas amount in microspheres.

where *X* is the radius of a reception platform of the detector, *x*,  $\Phi$  are the polar coordinates of the detector platform,  $r_0$  and *d* are the internal radius and wall thickness of a microsphere, *r*,  $\theta$ ,  $\varphi$  are the spherical coordinates for microsphere volume, *L* and *l* are the distances from the microsphere center up to the center of the reception platform of the detector and between any points of the microsphere and detector,  $l_i$ ,  $\mu_i$ are the length of *i*-th part of a way and factor of absorption in gas, microsphere wall, air, and a material of the detector window. The intensity of radiation generated in a microsphere wall is equal to

$$w = \frac{A_T n_T}{4\pi} \cdot \gamma \cdot n \int_0^{2\pi} d\varphi \int_0^{\pi} \cos \vartheta \cdot d\vartheta \int_0^{r_0} r^2 dr$$
$$\times \int_{E_c}^{E_m} f(E) dE \int_0^{R_c} \phi(E, x) \cdot \sigma(\varepsilon) \cdot dx.$$
(2)

Here  $A_T$  and  $n_T$  are a constant of decay and density of T particles in the microsphere,  $\gamma$  is the efficiency of fluorescence, *n* is the density of atoms of measured element,



Fig. 5. Spectrum of X-ray radiation of the target filled with T and argon.

Table 2. Definition of gas pressure in microspheres

	P (MPa)						
Number of microsphere	X-ray						
	K- series Si	K- series K	Optical measurement				
1	4.8	7.3	4.0				
2	3.1	3.6	2.3				

f(E) is the function of energy distribution of  $\beta$ -electrons,  $\phi(E, x)$  is the function considering elastic scattering of electrons and loss of its energy at movement in the environment (Reed, 1975; Pavlova *et al.*, 2000),  $\sigma(\varepsilon)$  are the cross-section of ionization from a corresponding electron shell (for example, K-shell) at collision of atom with  $\beta$ -electron having energy  $\varepsilon$ ;  $E_C$  and  $E_m$  are the boundary energy of ionization and maximal energy of  $\beta$ -electrons,  $R_e$  is the distance on which the electron energy decreases up to  $E_C$ .

In Table 2, results of calculation of gaseous DT (0.5:0.5) pressure for two microspheres on spectral data of K-lines of Si and K with the use of Eqs. (1) and (2) are presented. Values of parameters for calculation have been made (Reed, 1975; Pavlova *et al.*, 2000; Nemets & Hofmann, 1975; Mukhin, 1974; Krause, 1979). The values of pressure obtained experimentally by the method of optical interferometry (Veselov *et al.*, 1981) are here given.

From Table 2 it is visible that there is a significant difference in the calculated value of pressure from the value obtained by an optical method. Therefore, calculation on fundamental parameters can be used today only for an estimation of the value. For more exact determination, calibration of all measuring scheme as a whole is required. Comparison of calculated and experimental spectra of brake radiation has shown their coordination with accuracy of 30-40%.

The amount of Ar in microspheres containing T is based on the comparison of intensity of Ar characteristic peak and intensity of peak of a reference element (for example, Si), or continuous radiation in the power range of Ar peak. In both cases, the T amount almost does not influence the result. In the second case, for the definition of Ar pressure p in the microsphere, a simple formula (Weinstein & Weir, 1980) is used

$$p = C \cdot \frac{Q}{r_0},\tag{3}$$

where Q is the ratio of intensity of Ar line to intensity of continuous radiation in a spectral range of this line. Factor C is a constant in small limits of change of  $r_0$  and d, and at small amount of Ar. Its value can be defined in the calibration experiment. For use of Eq. (3) in a wider range of  $r_0$ , d, and p, it is necessary to remember that C is a function of these parameters and Eq. (3) becomes a nonlinear equation.

Except for the examples specified here, the spectral measurements of intensity of X-ray radiation can be used



**Fig. 6.** Spectrum of X-ray radiation of the shells filled with sulfur hexafluoride ( $SF_6$ ).

for studying solubility of T in various materials. Base expressions are Eqs. (1) and (2) with some changes.

#### 3.1.2. Targets without T

The spectrum of the microsphere filled with sulfur hexafluoride is presented in Figure 6. In spectrum, peaks of Si and K are a part of the glass microsphere, and sulfur is a part of the SF<sub>6</sub> molecule. Other peaks relate to the scattered radiation of the X-ray tube with a vanadium anode. It is visible that there is practically no background radiation.

The determination of gas amount in microsphere is made by comparison of intensity of characteristic radiation of the element, which is part gas, and a reference element of shell. The quantity of sulfur hexafluoride was defined by the intensity of the sulfur K-line. As a reference element, the Si in the glass shell is taken. The partial pressure of gas p in a target can be defined by expression

$$p = k(r_0, d, p) \cdot \frac{d}{r_0} \cdot \frac{S_g}{S_{Si}},\tag{4}$$

where  $S_g$  and  $S_{Si}$  are the areas under corresponding peaks of a gas element and Si, and d,  $r_0$  are the wall thickness and radius of shell, accordingly. The factor k is or in calibrating experiment or by calculation determined.

Fundamental equations for calculation k are equations for power of characteristic lines radiation of elements of gas  $S_g$ and glass  $S_{Si}$ . Division of  $S_g$  into  $S_{Si}$  gives Eq. (4) with factor

$$k = \left[1 + \frac{d}{r_0} + \frac{1}{3} \left(\frac{d}{r_0}\right)^2\right] \cdot \frac{3RT\rho_{gl}C_{Si}}{M_{Si}} \cdot \frac{\gamma_{gl}T_{gl}}{\gamma_g T_g} \cdot Q, \tag{5}$$

Q(r, d, p) is a complex function of radius, wall thickness of the shell, and gas pressure, and within limits of their change of 0–0.5 mm, 0–10 µm, and 0–10 atm., its value is in the range of 0.6–2.5.

	Para	ameters of	shell		Experiment			Calculation				
Gas	r <sub>0</sub> (mer	d m)	p (MPa)	Sg	S <sub>Si</sub> (pulse number)	q	k (MPa)	P <sub>g</sub> (p	<i>P<sub>Si</sub></i> pulse number	Q r/s)	k (MPa)	δ (%)
SF <sub>6</sub>	192.5	1.5	0.1	409	1505	0.27	47	0.57	0.35	0.613	46	2.6
Ar	189	1.2	0.09	273	379	0.72	22	0.98	0.38	0.39	20	10

Table 3. Results of comparison of calculated and experimental value of factor k

Results of the comparison of calculated and experimental values of the actor k are shown in Table 3. It is visible that the difference between them does not exceed 10%. The analysis of the influence of the measurement scheme parameters has shown that the main influence on the results of the measurement exerts the spectrum of exciting radiation. For example, carrying out during a measurement series, a change of the intensity or a full disappearance of the continuous spectrum of the exciting radiation a mistake of up to 20% may occur if these changes are not considered. Therefore, the stabilization of the spectrum of exciting radiation improves considerably reproducibility and accuracy of measurements.

#### 3.2. Measurement of Composition and Thickness of Flat and Profile Film Samples

The fullest and adequate description of properties of multilayered films, the combination of X-ray absorption, and fluorescence (RFA) techniques of the analysis can be given. They allow defining composition and mass thickness of layers of a film. The X-ray absorption method has the minimal mistake of measurement at greater film thickness while the RFA method gives the minimal mistake at small thickness (Fig. 7). Use of a probing beam of small diameter allows to carry out scanning on the area of the sample and to develop not destroying control of finished multi-layered and multistage targets.



Fig. 7. Dependence of a relative mistake of film thickness measurement by RFA (1) and absorptive (2) methods.

The scheme of film parameter measurements with use of both methods is shown in Figure 8. Radiation of X-ray tube 1 goes through diaphragm 7 on the secondary targets and excites in it probing radiation in the form of line spectra of the elements, which is a part of the secondary target. Probing radiation passes through collimator 5 and scanning diaphragm 6, through sample 4 and detector 3 and fluorescent radiation—to detector 3a. Enclosed by shaped lines area 8 containing scanning diaphragm and the sample should be evacuated down to an entrance window of the detector.

At definition of film thickness by RFA method, it used the comparison of intensity of an analytical line of measured film and the massive sample. At excitation of the film sample by characteristic radiation of a secondary target, the equation for definition of mass thickness of a film m may be given by

$$\frac{I_{m}^{i}}{I_{\infty}^{i}} = \frac{\sum_{r} \frac{I(E_{r})\tau_{i}(E_{r})\{1 - \exp\left[-K_{i}(E_{r})m\right]\}}{K_{i}(E_{r})}}{\sum_{r} \frac{I(E_{r})\tau_{i}(E_{r})}{K_{i}(E_{r})}},$$
(6)

where  $I_m^i$  and  $I_{\infty}^i$  are intensity of fluorescence of *i*-th element of a film and massive standard sample;  $I(E_r)$  is the intensity of quantum with energy  $E_r$  of characteristic radiation of secondary target material; *r* is the index of summation on all



Fig. 8. Scheme of measurement of film parameters. (1) the X-ray tube, (2) secondary target, (3, 3a) X-ray detectors, (4) sample, (5) collimator, (6) scanning diaphragm, (7) diaphragm.

lines of fluorescence of target material;  $\tau_i(E_r)$  is the section of photo effect for a quantum with energy  $E_r$  on atoms' *i*-th element;  $m = \rho \cdot d$  is the mass thickness of the sample;  $\rho$  is the density of the sample; *d* is the thickness of the sample;  $K_i(E_r)$  is the factor considering absorption in a material of film for a quantum of exciting radiation and fluorescence of chemical elements of a film, and also geometry of measurements;

$$K_i(E_r) = \frac{\mu(E_r)}{\sin\varphi} + \frac{\mu(E_i)}{\sin\psi};$$
(7)

 $\mu(E)$  is the mass factor of absorption in the sample for a quantum with energy E;  $\varphi$ ,  $\psi$  are the angles of sliding of exciting and registered radiation, accordingly. The bottom boundary of a range of film thickness values measured by method RFA is caused by a limit of detection of analytical peaks of film fluorescent radiation on a background substrate of a registered spectrum.

In X-ray absorption analysis, the dependence of transmission spectral factor of X-ray radiation from thickness and element structure of an investigated material is used, and at known element composition of a multi-layered film, its thickness is defined by means of the following equations set

$$\sum_{i=1}^{N} \mu_i(E_1) \cdot m_i = -\sin\varphi \cdot \ln\left[\frac{I(E_1)}{I_0(E_1)}\right]$$
$$\sum_{i=1}^{N} \mu_i(E_2) \cdot m_i = -\sin\varphi \cdot \ln\left[\frac{I(E_2)}{I_0(E_2)}\right], \quad (8)$$

$$\sum_{i=1}^{N} \mu_i(E_N) \cdot m_i = -\sin \varphi \cdot \ln \left[ \frac{I(E_N)}{I_0(E_N)} \right]$$

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where  $I(E_j)$  and  $I_0(E_j)$  are measured intensities of quantum with energy  $E_j$  past by sample and in absence of the sample, accordingly;  $\mu_i(E_j)$  are factors of loss of quantum with energy  $E_j$  in films consisting from *i*-th elements;  $m_i$  is the mass thickness of a film from *i*-th element. The solving of Eq. (8) gives thickness of each homoelementic film of which the multi-layered sample is made.

The analysis of profile films is carried out by a transparent X-ray beam with small cross-section size created by diaphragms, collimators, and X-ray lenses. Moreover, the use of X-ray charge-coupled camera with high resolution is possible.

The distribution of intensity of Pb film fluorescence with thickness of 5  $\mu$ m in the form of a strip with a width of 2 mm (the top curve) on an Al substrate by thickness 20  $\mu$ m in Cu mandrel (the bottom curve) on a surface coordinate is presented in Figure 9. Results of scanning by the method of local X-ray absorption analysis of the sample made of a Cu and Al film with a thickness of 5 and 20  $\mu$ m accordingly, are presented in Figure 10. The diameter of the scanning beam was 0.2 mm; the step of scanning is 0.1 mm.

The 4% inhomogeneity of the intensity of Pb L-band fluorescence in the scanning coordinate visible in Figure 9 is caused mainly by a surface roughness which makes  $0.1 - 0.2 \,\mu$ m. The heterogeneity of the intensity of the radiation transmitted to the Cu-Al sample with respect to the coordinate of scanning in Figure 10 does not exceed 1-2%.

#### 3.3. Conclusion for Measurement of Laser Fusion Targets Parameters by X-Ray Spectral Methods

Measurement of composition of the targets containing T is possible with the RFA method. The definition of the



Fig. 9. Intensity of fluorescence of a Pb film (1) and copper mandrel (2) in various points of scanning (excitation from side of Pb film).





**Fig. 10.** Intensity of characteristic radiation quantum of a secondary target past the Cu-Al sample depending on scanning coordinates. (1) target from germanium ( $E_{Ka} = 9.88 \text{ } \kappa \Im B$ ), (2) target from nickel ( $E_{Ka} = 7.47 \text{ } \kappa \Im B$ ).

amount of any element including T with use of only fundamental parameters gives a greater mistake and allows receiving only a qualitative value of the measured parameter. It is connected with difficulties of the adequate description of  $\beta$ -electron passage in the plural-component multi-layered environment. The use of reference elements or data about brake radiation allows improving the accuracy of measurements considerably.

Definition of composition of targets without T by means of RFA with use of a reference element allows receiving accuracy up to 10%. Application of monochromatic radiation for fluorescence excitation (secondary target) gives a reduction of the mistake of measurement by two to five times.

The complex application of RFA techniques and absorption analysis allows defining element composition and thickness of both single-layered and multi-layered films applied to micro targets manufacturing. By a choice of optimum energy of the quantum raying the sample or exciting fluorescence, it is possible to achieve the beautiful accuracy of optical thickness measurement for film samples in a wide range. The RFA method gives greater sensitivity and high accuracy for small thickness, and the absorption method has high accuracy at average and greater thickness. Their joint use allows to measure mass thickness of films in a range of  $10^{-6}$ –0.1 g/ cm<sup>2</sup> ( $10^{-3}$ –100 µm) with accuracy of 1-2%. The use of a probing exciting X-ray beam of small diameter or a beam with the slot-hole form allows developing scanning measurements on the sample area.

#### 4. RESEARCH OF STEEL SATURABILITY AND DESORPTION OF HEAVY HYDROGEN ISOTOPE (V. M. Izgorodin, A. P. Pepelyaev, Yu. V. Ignat'ev, E. Yu. Solomatina)

Lately, the method of registration of the secondary radiation (fluorescence) caused by  $\beta$ -electrons of T (Matsuyama *et al.*, 2002; Shmayda *et al.*, 2002), is being developed intensively and used for various purposes. The feature of the given

method is that the X-ray radiation generated in a thin  $(0.5-25 \ \mu\text{m})$  near-surface layer of a sample is registered. Actually, the attenuation factor of X-ray radiation with the energy concluded in a power range of T  $\beta$ -electrons is about 20000–400 cm<sup>-1</sup> (Nemetc & Hofman, 1975). Therefore, the fluorescence analysis is an effective method for studying absorption and desorption of T, and its thickness distribution in samples of various materials. The results of research can be assigned to all other isotopes of hydrogen and applied in many hydrogen technologies (hydrogen energetic, hydrogen engines, nuclear fusion, etc.), and at environment pollution research also.

#### 4.1. Experimental

We carried out experiments on saturability with T and kinetics of its desertion for some materials. Those materials were steel 30, stainless steel 12H18N10T, and also alloy EP543U-ID. Three sets of cylinder samples were made. Samples of one set were coated with Au of about 1  $\mu$ m thick. Samples with T-charging were conducted at temperature 250°C within 5 days in a special chamber filled with DT mix (0.5:0.5), up to a pressure of 0.4 MPa (the first set of samples) and 4 MPa (the second and third sets, the latter was coated with Au). In the samples of the first set aperture 8 mm in length and 2 mm in diameter was drilled along an axis of the cylinder.

After saturation relative activity monitoring of the samples was performed by means of a proportional gas counter, which registers energy integrated quantum intensity of the X-ray radiation exited by T  $\beta$ -electrons. Besides X-ray spectra of the samples were registered by means of Si (Li) detector. For an estimation of the absolute value of T concentration in near-surface layer of the samples hollow glass microspheres were filled with DT mix and spectra of their X-ray radiation were registered.

#### 4.2. Results

The quantity of impulses for the fixed interval of time received by an integrated gas counter for all the samples depending on time of storage at room temperature are presented in Figures 11 to 13. The figures indicate the monotonous decrease of sample activity. The reduction of the samples activity can be described as follows:

$$N = a + b \cdot \exp\left(-t/\tau_1\right) + c \cdot \exp\left(-t/\tau_2\right). \tag{9}$$

Values of *a*, *b*, *c*,  $\tau_1$ ,  $\tau_2$  are given in Table 4. We can conclude that the activity decline process can be divided into two processes—a galloping one with small time constant  $\tau_1$  and a slow one with big time constant  $\tau_2$ . We could not determine the value of  $\tau_2$  for the sample of steel 30 because of small observation time.

The dispersion of experimental point's values is caused by the impossibility to control the sample position relative to the detector precisely. In case of Au coating, the dispersion



Fig. 11. Impulse number dependence on time received by the counter (sample set 1).



Fig. 12. Impulse number dependence on time received by the counter (sample set 2).



Fig. 13. Impulse number dependence on time received by the counter (sample set 3).

Table 4. Factors values in Eq. (9)

Specimen		Specimen a		ecimen a b c		$\tau_1 \text{ (days)}$	$\tau_2$ (days)
Steel30	8500	7391	-	260	~		
12H18N10T	1970	4580	4840	60	930		
EP543U-ID	1000	4590	2360	24	204		
Steel30	9380	6990	16770	44	604		
12H18N10T	3380	6750	13630	23	245		
EP543U-ID	3920	5050	14560	15	150		
Steel30	4210	2110	10470	30	880		
12H18N10T	3900	1660	11020	32	680		
EP543U-ID	3300	1710	10580	21	820		
	Steel30 12H18N10T EP543U-ID Steel30 12H18N10T EP543U-ID Steel30 12H18N10T EP543U-ID	specimen         a           Steel30         8500           12H18N10T         1970           EP543U-ID         1000           Steel30         9380           12H18N10T         3380           EP543U-ID         3920           Steel30         4210           12H18N10T         3900           EP543U-ID         3000	a         b           Steel30         8500         7391           12H18N10T         1970         4580           EP543U-ID         1000         4590           Steel30         9380         6990           12H18N10T         3380         6750           EP543U-ID         3920         5050           Steel30         4210         2110           12H18N10T         3900         1660           EP543U-ID         3300         1710	a         b         c           Steel30         8500         7391         -           12H18N10T         1970         4580         4840           EP543U-ID         1000         4590         2360           Steel30         9380         6990         16770           12H18N10T         3380         6750         13630           EP543U-ID         3920         5050         14560           Steel30         4210         2110         10470           12H18N10T         3900         1660         11020           EP543U-ID         3300         1710         10580	b         c         τ1 (days)           Steel30         8500         7391         -         260           12H18N10T         1970         4580         4840         60           EP543U-ID         1000         4590         2360         24           Steel30         9380         6990         16770         44           12H18N10T         3380         6750         13630         23           EP543U-ID         3920         5050         14560         15           Steel30         4210         2110         10470         30           12H18N10T         3900         1660         11020         32           EP543U-ID         3300         1710         10580         21		

increases because of Au layer imperfection and corrosion plays a great role in case of steel 30.

It is clear from the table that the time constants of activity decay  $\tau_1$  and  $\tau_2$  decrease from steel 30 to alloy EP543U-ID for not coated samples. For the samples saturated with T at 0.4 MPa,  $\tau_1$  and  $\tau_2$  are greater than for the sample coated at 4 MPa. In case of the samples coated with Au, the constants of time  $\tau_1$  and  $\tau_2$  weakly differ from each other.

Energy spectra of samples radiation of sets 2 and 3 (covered with Au and uncovered) are shown in Figures 14 and 15. One can see from the figures that it is possible to identify a lot of elements, which form steels: for steel 30 it is Fe, Mn; for 12H18N10T it is Fe, Cr, Ni, Ti; for alloy EP543U-ID it is Fe, Cr, Ni, Ti, Mo. In the spectra of sample set 3, the powerful peak of Au M-band is visible as well as steel elements radiation weakened by the Au layer. All spectra are accompanied by a continuous background of braking radiation.

#### 4.3. Discussion of Results

Using the method of fundamental parameters at fluorescence analysis (Mantler, 1987), it is possible to estimate T density in radiating area of the sample depending on spectral lines intensity and known steel composition. The intensity of the analyzed element peak of the cylindrical sample, which lateral surface is parallel to a reception platform of the detector, is possible to express in the form of

$$\dot{N} = \frac{1}{4\pi} \int_{0}^{2\pi} d\Phi \int_{0}^{X} x \cdot dx \int_{-l0/2}^{l0/2} dy \int_{\alpha 1}^{\alpha 2} d\alpha \int_{0}^{r_{0}} w \cdot \frac{L - r\cos\alpha}{l^{3}}$$
$$\times \exp\left(-\sum \mu_{i} l_{i}\right) \cdot rdr, \qquad (10)$$

where *X* is the radius of a reception platform of the detector; *x*,  $\Phi$  are the polar coordinates of the detector platform,  $r_0$ ,  $l_0$  are the radius and length of the cylinder; *r*, *y*,  $\alpha$  are cylindrical coordinates of the sample,  $\alpha_1$  and  $\alpha_2$  are boundary corners of detector visibility from a point of the sample surface, *L* and *l* are the distances from the center of the sample up to the



Fig. 14. Radiation power spectrum of sample set 2 (not covered).

center of a reception platform of the detector, and from any point of the sample up to a point of the detector;  $l_i$ ,  $\mu_I$  are the length of *i*-th segment of a way and factor of absorption in the sample, air, and window material of the detector. Intensity of radiation generated in the sample is equal to

$$w = -A_T \cdot n_T \cdot \gamma \cdot n \int_{E_C}^{E_m} f(E) \cdot dE \int_{E_C}^{E} \frac{dR}{d\varepsilon} \cdot \sigma(\varepsilon) \cdot d\varepsilon.$$
(11)

Here  $A_T$  and  $n_T$  are the decay constant and density of T particles in the sample,  $\gamma$  is the fluorescence efficiency, *n* is the density of atoms of an analyzed element, f(E) is the function of  $\beta$ -electrons energy distribution,  $\sigma(\varepsilon)$  is the ionization cross-section of a corresponding shell (for example, K-shell) during atom collision with  $\beta$ -electrons with energy  $\varepsilon$ ;  $E_C$  and  $E_m$  are the boundary energy of ionization and maximal energy of  $\beta$ -electrons,  $d\varepsilon/dR$  is the energy loss speed of electron during its passage in the sample.

Similar calculations for two microspheres filled with DT mix were made in order to check these calculations' correctness. Eqs. (1) and (2) were used. The results of the calculations were compared with values of gas pressure obtained by the method of optical interferometry (Veselov *et al.*,



Fig. 15. Radiation power spectrum of sample set 3 (covered).

1981). Results of calculations and measurements by an optical method are presented in Table 2. Pressure values are calculated using intensity of Si and K peaks. It is visible that the value of pressure obtained by an X-ray method is 1.2–1.8 times larger than the optical one. Therefore, data found by an X-ray method for concentration of T may be accepted to be correct by a factor of two.

A calculated value of T concentration in the samples is a result of intensities of separate peaks elements in spectra and thickness of near-surface layer from which radiation was registered (Table 5). The thickness of this layer is understood as distance where radiation with the given energy is weakened three times. Thus, the T concentration presented in the table is the near-surface areas' average value. Literary data about  $\beta$ -electrons energy distribution runs and factor of electron elastic dispersion, ionization sections and fluorescence efficiency, and also spectral factors of X-ray radiation weakening from (Nemetc & Hofman, 1975; Reed, 1979; Pavlova *et al.*, 2000; Mukhin, 1974; Krause, 1979) were used for the calculations.

Table 5 shows that the dispersion of T concentration calculated using characteristic peaks of various elements is not so great. Average values of T concentration for different samples and average scattering are given in Table 6.

Table 6 indicates that the T concentration in the nearsurface layer is 1.8 times greater for steel 30 at increase in saturation pressure from 0.4 MPa up to 4 MPa. At the same time, the T concentration increases 2.4 times in 12H18N10T and 10 times in alloy EP543U-ID at the same increase in saturation pressure.

**Table 5.** Calculated values of T concentration in the samples  $(10^{18} \text{ sm}^{-3})$ 

Material	Element	Set 1	Set 2	Determination region (µm)
Steel 30	Fe	9.6	17	21
12H18N10T	Ti	4.9	16	8
	Cr	9.8	24	11
	Fe	8.1	18	9
	Ni	6.3	11	43
Alloy 543U-ID	Ti	2.7	27	6
5	Cr	2.4	24	9
	Fe	2.3	25	9
	Ni	1.9	22	6

**Table 6.** Average T concentration values for different samples  $(10^{18} \text{ cm}^{-3})$ 

Material	Set 1	Set 2	
Steel 30	9.6	17	
12H18N10T Alloy EP543U-ID	$\begin{array}{c} 7.2 \pm 1.8 \; (25\%) \\ 2.3 \pm 0.3 \; (12\%) \end{array}$	$17 \pm 4.6 (27\%) \\ 24 \pm 1.8 (7\%)$	



Fig. 16. Experimental and calculated spectra of  $\beta$ -electrons braking radiation in microsphere.

As the gas-discharge counter was energy integrated, the quantity of counted impulses is defined by both characteristic and a continuous spectrum of radiation. Calculated and experimental spectra of braking radiation of one of the microspheres are given in Figures 15 and 16.

For the calculation of the braking radiation intensity generated in microsphere, Eq. (1) was used. In this equation, w was defined as follows

$$w = \frac{A_T n_T}{4\pi} \int_0^{2\pi} d\phi \int_0^{\pi} \cos \vartheta \cdot d\vartheta \int_0^{r_0} \frac{r^2 dr}{R^2} \int_{E_c}^{E_m} f(E)$$
$$\times \int_0^{R_e} \phi(E, x) \cdot \sum_i n_i \left( k \frac{d\sigma_i(k, \varepsilon)}{dk} \right) \cdot dx, \qquad (12)$$

where k is energy of X-ray quantum;  $n_i$  and  $d\sigma_i(k, \varepsilon)/dk$  are atoms density and cross-section of X-ray quantum radiation with energy k at braking of electron with energy  $\varepsilon$  for an *i*-th element of substance (Pratt *et al.*, 1977). Here, the consent between calculated and registered spectrum is good enough.

Thus, course of time dependence of samples activity registered by the gas-discharge counter is possible to use for quantitative estimations of absolute value of T concentration and to estimate this value at the moment of time right after saturation.

Results of such estimation are shown in Table 7. Here, values of T concentration from Table 6 as reference values and time dependences of activity are shown in Figures 11

**Table 7.** Estimated concentration of T in time right after saturation  $(10^{18} \text{ cm}^{-3})$ 

Material	Set 1	Set 2
Steel 30	13	23
12H18N10T	13	33
Allov 543U-ID	9	51

and 12. The registration of spectra was carried out for samples from the first set on day 191 and the second sets on day117 after saturation. Figures 11 and 12 and Table 6 show that the quantity of the counted activity decreased for steel 30 about 1.3 to 1.4 times, for 12H18N10T about 1.8 to two times, and for alloy EP543U-ID about four and two times in set 1 and set 2, accordingly, for these time intervals.

### 4.4. Conclusion for "Research of Steel Saturability and Desorption of Heavy Hydrogen Isotope"

The kinetics of T desorption from samples of three types of steel: steel 30, stainless steel 12H18N10T, and alloy experimentally EP543U-ID was investigated. The desorption-character can be described by two declining exponential functions with small and large time constant  $\tau_1$  and  $\tau_2$ . When the saturation pressure was 0.4 MPa, time constants  $\tau_1$  and  $\tau_2$  came to 260 and  $\infty$  days for steel 30, 60, and 930 days for stainless steel 12H18N10T and 25 and 200 days for alloy EP543U-ID. With increase of saturation pressure up to 4 MPa, the time constants became equal to 45 and 600 days for steel 30, 25, and 250 days for stainless steel 12H18N10T and 15 and 150 for alloy EP543U-ID. For samples, coated with Au  $\tau_1$  and  $\tau_2$  are 20–30 and 700–900 days, respectively, for all materials. Desorption in that case is determined by gas diffusion through Au.

It is shown that the use of fundamental parameters method at fluorescence analysis allows performing measurements of T concentration and studying of its evolution in the nearsurface layer in various saturated samples with change of time. Specification of fundamental parameters values, careful fixing of measurement geometry, and development of the account of non elastic electron dispersion in a calculation technique, will allow improving the accuracy of measurements considerably. Computation estimations of T concentration absolute value in near-surface layer of samples with use of experimental peak intensities of characteristic radiation gave the following values:  $13 \times 10^{18}$ ,  $13 \times 10^{18}$  and  $9 \times 10^{18}$  cm<sup>-3</sup> at the moment of time right after saturation at pressure 0.4 MPa, and  $23 \times 10^{18}$ ,  $33 \times 10^{18}$  M  $51 \times 10^{18}$  cm<sup>-3</sup> at pressure 4 MPa for steel 30, stainless steel 12H18N10T and alloy EP543U-ID accordingly.

#### 5. TARGETS FOR STUDYING THE EQUATION-OF-STATE (A. P. Balyaev, A. V. Veselov, S. G. Garanin, A. G. Golubinsky, V. M. Izgorodin, E. Yu. Solomatina, I. N. Cherkesova, V. V. Chulkov)

The target for studying the equations of state target consists of an Al base plate with a thickness of  $50-100 \mu m$ , two Al layers in the form of steps, and an investigated material. Steps are put on one of the surfaces of a base plate and divided by an interval of  $50-150 \mu m$  (Fig. 17). The height of the steps is  $4-10 \mu m$ . Investigated materials are Pb or Cu. In some cases, the side of a target with steps is coated with  $0.3-0.6 \mu m$  layer of Al. This is done for alignment of intensity of the light radiation arising at an output of a



Fig. 17. The scheme of EOS target. On the left is an arrangement of layers, on the right is disposition of a target on the holder: (1) the holder, (2) target, (3) aperture for adjustment.

#### Target Technology in RFNC-VNIIEF (Sarov)

shock wave on a target surface. Sometimes the other side of the target, which a laser impulse influenced, is covered with a layer of polyparaxylilene (parylene) with thickness of  $8-10 \mu m$ . All this assembly is placed on the flat steel holder opposite to a diagnostic aperture with a diameter of 1 mm. The additional aperture for the target adjustment with diameter of  $30-40 \mu m$  is done on the holder that is 6 mm from the diagnostic aperture.

The accuracy of the hock wave velocity measurement is defined in particular by the quality of target manufacturing, therefore, accuracy requirements for working layers in equations of state targets are high (Koenig *et al.*, 1995). The nonuniformity of layers thicknesses should be <1% and a base plate of <0.4%. The target deviation is formed from surface roughness, nonuniformity of thicknesses of a base plate, and step layers, and also inaccuracy for these sizes measurements.

#### 5.1. Manufacturing of a Base Plate and Steps by Pressing Method

Originally, the base plate was an industrial Al foil with the thickness of  $50-100 \mu m$  and made by a rolling method. On the surface of the foil were characteristic inhomogeneities in the form of strips. The amplitude of surface inhomogeneity runs up to  $0.5-0.6 \mu m$  (Fig. 18a). Besides a Cu industrial foil with a thickness of  $30-50 \mu m$  was used. Its surface inhomogeneity was two to three times less. Steps of investigated metals were deposited by resistive evaporation in vacuum, or magnetron sputtering. At Al steps deposition by magnetron sputtering, the surface of a layer was

opaque, and did not give a fringe pattern. Resistive sputtered Cu films were the best (their surface repeated a structure of a substrate surface almost precisely). Resistive coated Pb films were rough.

An industrial foil pressing has been applied to receive smoother surface of a base plate (Krasnikov, 1976; Gunn *et al.*, 1974). For this work, a foil piece was pressed out between polished surfaces of steel plates. The amplitude of foil surface inhomogeneity was  $0.1-0.2 \mu m$  (Fig. 18b).

Moreover, we attempted to make steps by foil pressing to reduce the roughness and intermediate layer width. For that, a good matrix was necessary. Attempts to make matrixes of high quality have shown that the best results turns out to be an assembly of a matrix from the polished details. By development of a matrix and mirrors design, pressing conditions have allowed to receive at press of Al foil steps with height up to  $10-20 \,\mu\text{m}$  and width of transitive layer  $1-3 \,\mu\text{m}$ , with small roughness (~0.01  $\mu$ m), and good parallelism of surfaces. Technological advancement enables to make the adjustable step size. The 3D image of a step made with scanning interferometer is presented in Figure 19.

The variant two steps production by means of one matrix has been tested too. The prototype of such a matrix has been made. It enables to assemble three variants of steps. 3D images of steps on Al foil made by means of such a matrix are presented in Figure 20. The two-stage targets on a Cu foil have been received too.

The further update of a matrix design from the point of view of rigidity, convenience of adjustment of the demanded



Fig. 18. Microinterferograms of base plate surface ((a) before pressing, (b) after pressing).



Fig. 19. 3D image of a step surface with height 11.2  $\mu m$  .



Fig. 20. 3D images of steps on Al foil made by means of two-level matrixes, the width of a mean plane is 100  $\mu$ m, steps heights are: (a) 9.7 and 16.5  $\mu$ m, (b) 7.4 and 6  $\mu$ m, (c) 3 and 10  $\mu$ m.

height of steps, durability will allow to receive good repeatability of target parameters and an opportunity of their mass manufacturing.

### 5.2. Manufacturing of Metal Steps by Vacuum Evaporation

Manufacturing of metal step layers was carried out by resistive evaporation in vacuum through a mask, close-fitting to a substrates surface. In this case, the sharpest step border is obtained, and the width of a transitive zone is  $5-15 \,\mu\text{m}$ . At magnetron sputtering, this size was in two to four times more.

#### 5.2.1. Manufacturing of Cu and Pb Step Layers

The resistive sputtering from tape heater was used for Cu and Pb steps deposition. The height of steps was up to 10  $\mu$ m. It is possible to manufacture multi-stage (3–4 steps) targets from one material by mask shifting. The interferogram received on micro-interferometer MII-4 two-step target Al/Cu (height of 7 and 5  $\mu$ m) on a Cu substrate with thickness of 40  $\mu$ m is shown in Figure 21. The width of a transitive zone of steps does not exceed 10  $\mu$ m.

The condition of a Cu layer surface is good. The Pb surface becomes very rough at a thickness above  $0.5 \,\mu\text{m}$ . The surface profiles of a Cu layer and the Pb layer are represented in Figure 22. They were measured with an atomic



Fig. 21. Interferogram received on micro-interferometer MII-4 of the target with Al and Cu steps (heights 7 and 5  $\mu$ m ) on a Cu base plate (thickness 40  $\mu$ m ).



Fig. 22. Surface structures of the copper film (a) and the Pb film (b), measured with an AFM.

force microscope Solver P47. The root-mean-square roughness is about 80 nm for Pb, and 10 nm for Cu. The cross-section size of grains was about 30 nm for Cu and 300-400 nm for Pb.

#### 5.2.2. Al Layers Manufacturing

Films of Al in a range of thickness up to 10  $\mu$ m were produced by sputtering in vacuum on installation VUP-5 with oil-free pumping and working vacuum about 10<sup>-3</sup> Pa. Sputtering was fulfilled by means of an evaporator in which the pyrolitic boron nitride crucible was used. Thermometry of crucible bottom and automatic temperature level control were realized during sputtering. The working temperature of a crucible bottom was preset to a range of 1150–1200°C with accuracy  $\pm 2^{\circ}$ C. Clearing of substrate surfaces and chamber walls before film deposition were realized by ionic etching in an Ar atmosphere, warming up and prolonged pumping.

In these conditions, it was possible to produce thick Al film much more smooth, than at magnetron sputtering. The interferogram of surface of 12.6  $\mu$ m Al film is represented in Figure 23. It has been deposited on a glass substrate and



Fig. 23. Interference fringes received by micro-interferometer MII-4 on a surface of an Al layer with thickness12.6  $\mu$ m deposited on a glass substrate (a) and polished NaCl substrate (b).

a substrate from polished NaCl. A presence of enough precise interference fringes enables to prove that the amplitude of a roughness does not exceed 0.1  $\mu$ m. Measurements of these films roughness with scanning white light interferometric profilometer are received in both cases about identical values of 0.05–0.06  $\mu$ m.

Measurement of a film roughness with atomic force microscope (AFM) has given 70 nm for average amplitude of surface roughness at the cross-section size of grains of 20 nm. The structure of this film surface measured with AFM is represented in Figure 24.

Single- and multi-stage targets were made using the crucible evaporator. The first step was deposited on a polished Al base plate with a thickness of 50  $\mu$ m, and then consecutively two to three steps of Al were deposited with shift of a precision mask on 50–100  $\mu$ m. The example of resulting surface profiles is represented in Figure 25.



Fig. 24. Surface profile of an aluminum layer with thickness  $12.6 \ \mu m$  on NaCl substrate, measured with AFM.



Fig. 25. A three-stage target manufactured by crucible evaporating aluminum with consequent precise displacement of mask.

Measurement of geometrical parameters for two-step (Al/Cu) targets by means of optical interference profilometer have shown that the characteristic sizes of targets are: step height (up to 10  $\mu$ m) with a surface roughness for Al, 0.05–0.2  $\mu$ m, for copper, 0.02–0.04  $\mu$ m, distance between steps, 50–150  $\mu$ m, and width of a transitive zone, 3–5  $\mu$ m. The surface roughness of a Cu layer is almost completely defined by the surface roughness of a base Al plate.

#### 5.3. Development of Optical Measurement of Step Height and Layer Roughness

Originally, the thickness of a layer and an estimation of a surface roughness were realized with Linnik microinterferometer MII-4. The layer thickness of opaque materials was defined by interference fringes shift at simultaneous supervision of a substrate surface and the deposited layer surface. For a continuous film, it was necessary to do a scratch to observe the substrate. If the film was transparent with a known refraction index, its thickness could be measured without presence of a sharp step or a scratch. The shift of fringes was defined visually, and the precision of the measurement could be not worse than 0.05  $\mu$ m at high quality of fringe pattern.

However, micro-interferometer MII-4 was developed in 1937, and ever since, these methods of surface structure measurement advanced forward. The developments in computer technologies, in charge-coupled-device-camera manufacturing, and the achievements in the exact mechanics have enabled to raise considerably the precision of measurements, their visualization, to be released from usual "operator's error" at measurement of surface imperfections. Scanning white light interferometric profilers provides vertical resolution <0.1 nm with the lateral resolution 0.43 mcm.

In our laboratory, we attempted to modernize the microinterferometer by adding on a model for scanning white light interferometer (Deck & de Groot, 1994; Sinclair *et al.*, 2005). The micro-interferometer MII-4 has been taken as a basis for the given device. A number of basic constructive changes have been made. Automatic moving of the optical system with a step of 0.01  $\mu$ m has been introduced, that has permitted to apply the scanning interferometric techniques (de Groot & Colonna de Lega, 2004).

This variant was described previously when talking about not destroying control of the structure of spherical shell surfaces. By means of the described device, 3D images of surface structures are shown in Figures 19, 20, 25. Now, the device allows making measurements of step height with accuracy of  $\pm 0.1 \,\mu\text{m}$  and measurements of a roughness with accuracy of  $\pm 0.01 \,\mu\text{m}$  at the resolution in a crosssection direction  $0.5-1 \,\mu\text{m}$ .

#### 5.4. Measurements of Film Thickness by X-Ray Spectral Methods

The combination of X-ray absorption and fluorescence (RFA) techniques of the analysis can give the fullest and

adequate description of properties of multi-layered film samples. They permit to define the composition and optical thickness of the films. The X-ray absorption method has the minimal measurement error at greater thickness of film samples (more than  $1-3 \mu$ m), while the X-ray fluorescent method gives the minimal error at smaller thickness. Furthermore, the fluorescent method enables to make the element analysis in multi-layered film samples. Use of a probing beam with small diameter enables to carry out scanning on the area of the sample and to develop not destroying control of finished multi-layered and multistage targets (Vasin *et al.*, 2007). As an illustration, there is a fluorescence intensity distribution on a surface of Pb film (the top curve) on Al substrate in Cu mandrel (the bottom curve) in Figure 26. Diameter of a scanning beam is 0.2 mm.

In Figure 27, there are presented the results of scanning by a local X-ray absorption method for sample made from Pb and Al films. The step of scanning is 0.1 mm. Evidently, uniformity of film thickness on surface region  $\sim 1$  mm is within the limits of 1-5%.

The combined application of X-ray techniques of fluorescence and absorption analysis permits to measure optical films thickness in a range of  $6 \times 10^{-7}$ –0.1 g/cm<sup>2</sup> ( $6 \times 10^{-4}$ –100 µm) with accuracy of 1–2% (Vasin *et al.*, 2007).



**Fig. 26.** Fluorescence intensity of Pb film (the top curve) and copper mandrel (the bottom curve) in various points of scanning (excitation from side of Al film).



**Fig. 27.** Intensity of characteristic radiation of a secondary target passed through the Pb-Al sample versus scanning coordinate. The bottom curve – a nickel target ( $E_{Ka} = 7.472$  KeV), the top curve is a zirconium target ( $E_{Ka} = 15.746$  KeV).

#### 5.5. Measurement of Film Thickness with Alpha-Spectrometry Method

Determination of layers and films thicknesses (self-supported and on a substrate) is carried out without destruction of the sample by measurement of energy losses in the sample for passing  $\alpha$ -particles (Bethe & Ashkin, 1953). At measurements by this method, the investigated sample is located in the vacuum chamber, where it is irradiated by  $\alpha$ -particles of Pu-239 source, which spectrum is registered by the Si surface-barrier detector. The thickness is measured on the displacement value of the peak center for  $\alpha$ -particles (5155.4 keV energy). To control the spectrometer path characteristics, the detector is in addition highlighted (sideways, passing the sample) by Pu-238  $\alpha$ -particles with energy of 5456.5 and 5499.2 keV. The method provides reliable registration of displacement from 15 keV.

Instrumental spectra are represented at thickness measurements for different Cu films in Figure 28. It is clear from the figure that there is a tendency of line width increase of  $\alpha$ -particles' power spectrum with film thickness increase. This effect for a case of a Cu film is caused mainly by straggling. However, in case of a Pb film, the roughness of a surface gives the appreciable contribution in broadening of the displaced peaks.

Thus, by this manner, it is possible to accept the ranges of thickness registration of  $0.1-15 \,\mu\text{m}$  for Al,  $0.01-8 \,\mu\text{m}$  for Cu, and  $0.05-12 \,\mu\text{m}$  for Pb. The measurement error is 4-5%. Unlikely, the measurement accuracy can be raised appreciably. However, received instrumental spectra enable to make estimations of the sample uniformity on thickness, density and composition, and also to define the presence of

cracks and chips. All these imperfections and damages Pb to a spectrum line spreading and to occurrence of additional peaks.

#### 5.6. Conclusion for Targets for Studying the Equation-of-State

Now, the following methods are applied for equation of state targets manufacturing: (1) pressing of industrial Al foil has enabled to reduce the height of the surface roughness from  $0.5-0.7 \mu m$  down to  $0.1-0.2 \mu m$ ; (2) surface polishing for Cu industrial foil permits to receive base plates with a surface roughness  $0.05 \,\mu\text{m}$ ; (3) the pressing method has enabled to produce single- and multi-stage targets from a homogeneous material, Al or Cu, with surface roughness less than  $0.05-0.1 \,\mu\text{m}$ , and width of a transition zone for steps no more than 3 µm; (4) the vacuum sputtering method at corresponding processing of the base plate surface gives the possibility to make Cu and Pb steps with a root-mean-square surface roughness of 10 nm and 80 nm, accordingly, and with transitive zone width no more than 10 µm. The average lateral dimension of roughness is 30 nm for Cu and 300-400 nm for Pb; (5) we succeeded in sputtering deposition films of Al with 10 µm thickness and RMS surface roughness of 50-70 nm, using the crucible evaporator and corresponding substrate processing.

Developments of measuring methods enable to register the parameters of targets with sufficient accuracy: (1) using of industrial white light optical interferometer MII-4 permits to measure a step height up to  $10 \,\mu\text{m}$  with an accuracy of 0.05  $\mu\text{m}$  and with lateral resolution of about 0.5  $\mu\text{m}$ . On the basis of this device, the scanning white light interferometer



Fig. 28. Instrumental energy spectra of alpha particles at copper film measurement (FWHM is line width on half-height).

has been created. It enables to receive the 3D image of surface structure in the  $150 \times 190 \text{ mcm}^2$  field-of-view and to measure a step height up to 50 µm with an accuracy of 100 nm; (2) development of X-ray spectral methods has enabled to measure film thickness with accuracy of 1-2% and by scanning to build a one-dimensional profile of layer thickness for films with various element structure; (3)  $\tau \eta \epsilon \alpha$ -spectrometry method permits to receive a qualitative value of film thickness heterogeneity on the chosen area of a target; (4)  $\tau \eta \epsilon$  use of an industrial atomic-power microscope enables to receive a 3D image of a film surface with accuracy of 0.1 nm and the lateral resolution of 1-10 nm. Measuring of the surface with a height step up to 5 µm is possible.

# 6. CONDENSED GASES JET GENERATOR FOR NANOLITHOGRAPHY (F. M. Abzaev, A. V. Bessarab, V. G. Gogolev, A. G. Golubinsky, V. M. Izgorodin, D. A. Irinichev, G. P. Nikolaev, I. M. Rojz, D. Ju. Fenoshin)

Development of lithographic devices using super hard ultraviolet radiation (nanolithograph) is carrying out in many micro-electronics centers at present. A source of such radiation is the laser plasma. The most preferable is the plasma activated in condensed noble gases, especially in Xe (Diefendorff, 2000; Hansson *et al.*, 2002).

The jet of condensed Xe as a source of ultraviolet radiation possesses some advantages. They are a rich spectrum of radiation in demanded area with the large efficiency of transformation of laser radiation and absence of contamination of a surface of optical details of nanolithograph during its work. However, at use of Xe there are also a number of problems. The main things from them are high cost and need of powerful system of evacuation.

#### 6.1. Theory and Estimations

Functionally, the jet generator consists of four basic parts: the pipeline bringing gas, the condenser, the store of liquid Xe, and the nozzle (Fig. 29).

Gas moves in the condenser on the pipeline and is preliminary cooled. Walls of the condenser are at constant temperature  $T_w$ . In the condenser, gas is cooled up to the temperature close to the wall temperature and condensed, if  $T_w$  is below the temperature of condensation  $T_K$ . In the store, the homogeneous velocity field of a liquid is formed on an input in nozzle, therefore the stability of the outflow in direction is provided. The expiration of a liquid occurs in the evacuated chamber.

# 6.1.1. Calculation of Parameters of Flow and Condensation

For the calculation of the mean of cross-section flow parameters in the jet generator, the equations for velocity, temperatures, pressure and the consumption of a liquid and gas, and also for the area of cross-section of a liquid phase were



Fig. 29. The block diagram of the jet generator. (1) the pipeline, (2) condenser, (3) store of liquefied gas, (4) nozzle, (5) jet.

used. A basis of the equations was the conditions of continuity and conservation of energy and impulse. The flow was supposed to be one-dimensional and stationary. At transition from a mode of only gas flow to a mode of gas-liquid flow, it was supposed that at excess of pressure above the pressure of condensation, the liquation of gas part occurs in steps with conservation of weight and energy (Samoilovich, 1990; Altshul *et al.*, 1987).

Results of calculations of gas flow parameters without condensation (pressure, velocity and temperature of gas, and also its density, Mach number and the mass consumption) are presented in Table 8. There the temperature of a condenser wall becomes equal to 80 K, and the temperature of the pipeline linearly decreases with distance from 300 K down to 80 K. A nozzle was thin enough (a length was much more less then a diameter).

It is clear that the Mach number is close to 1 on the exit of the nozzle, and the mass consumption is constant. It is the confirmation of calculations' correctness.

Dependences of the liquid consumption and gas velocity on distance in the condenser are resulted at liquation presence in Figure 30 as a sample of calculation for the case  $T_w < T_K$ .

## 6.1.2. Problem of Liquid Boiling-up in a Jet at Pressure Reduction

At movement of a liquid in the nozzle channel, the pressure decreases from initial pressure  $p_0$ , and reaches the value of the saturated vapor pressure  $p_s$  on the length of the way *l*. At the further movement, the pressure continues to decrease, and the liquid becomes overheated. As a result, there are

Number part	p (MPa)	u (cm/s)	T (K)	$\rho \; (g/cm^3)$	М	<i>ṁ</i> (g/s)
Initial value	0.1	88	300	0.0016	0.0027	0.00071
End of pipeline	0.0999696	25.6	87.2	0.0052	0.0015	0.00071
Begin of condenser	0.0999696	25.6	87.2	0.0052	0.0014	0.00071
End of condenser	0.0999693	23.6	80.5	0.006	0.0014	0.00071
Begin the store	0.0999693	1.026	80.5	0.006	0.00006	0.00071
End of the store	0.0999693	1.026	80.5	0.006	0.00006	0.00071
Enter from nozzle	0.0461728	14382	59	0.0049	0.89	0.00071

Table 8. Results of gas flow parameters calculations without condensation

conditions for fast growth of bubbles on small solid inclusions, and the liquid boils in all the volume (Frenkel, 1975).

The minimal radius of bubble is the radius of the nuclear particle formed by molecules of gaseous impurity ( $H_2O$ ,  $CO_2$ ). The radius of the nuclear particle grows constantly due to sticking to it of impurity molecules. Then in absence of solid particles of an impurity, at the beginning of condensation, the condition of not boiling-up of the



**Fig. 30.** Dependences of the liquid consumption (**a**) and gas velocity (**b**) on distance in the condenser. The curves 1, 2, 3, 4, 5 correspond to temperature of condenser wall of 160, 155, 150, 145 and 140 K.

liquid can be written in the form of

$$\frac{n_d}{n_{Xe}} < \frac{16\sigma}{p_S\sqrt{Dt}},\tag{13}$$

where  $n_d/n_{Xe}$  is the atomic concentration of impurity,  $\sigma$  is the factor of surface tension, *D* is the coefficient of molecule diffusion of an impurity in liquid Xe, and *t* is the time interval from the beginning of condensation. For the nozzle unit working temperature of 170 *K* and the time from the beginning of condensation of 10 min, an estimation gives the maximal maintenance of impurity 0.4 at.%.

If the liquid leaves the channel in vacuum before volumetric boiling-up, then after an output of a jet in vacuum, a strong evaporation begins on the surface. It leads to cooling of a liquid and suppression of volumetric boiling-up. Further, the liquid moves in the form of an integral jet that gradually solidifies. If the growth of bubbles occurs to the big velocity then at an output in vacuum the jet is sprayed in the form of vapor-liquid suspension losing the integrity.

Therefore, to provide the outflow of a jet without spraying it is necessary: (1) to work with well cleared gas; (2) at possibly low temperature; (3) at large velocity of the liquid outflow; (4) at probably smaller length of nozzle capillary.

#### 6.1.3. Problem of Jet Stability

The development of the instability of a jet direction is connected with turbulence. It arises mainly at excess of the Reynolds number over the critical value equal to 2000. Moreover, turbulence can be initiated by formation of large bubbles, gas cavities on capillary wall at liquid boiling-up and solid microparticles. The greatest velocity of liquid movement is in a capillary. For short capillaries, jet instability arises mainly because of the presence of the solid microparticles that have adhered to the wall of the nozzle.

#### 6.1.4. Behavior of a Jet in Vacuum

At an output of a jet in vacuum, it is cooled because of intensive evaporation of a liquid from its surface. The process of cooling is described by the equation of balance of power flow in cylindrical jet with radius *a*:

$$\frac{a\rho C_V}{2} \cdot \frac{dT}{dt} = -LG + Q_R + Q_T + Q_S. \tag{14}$$



Fig. 31. Dependences of average temperature (a) and relative radius (b) of a jet on time for initial diameters 20  $\mu$ m (1) and 40  $\mu$ m (2).

Here,  $\rho$  is the density of the liquid,  $C_V$  is the heat capacity, T is the average of radial temperature, L is the heat of evaporation, G is the speed of evaporation of liquid,  $Q_R$  is the heat leakage due to radiation of chamber wall,  $Q_T$ 

is the heat leakage due to heat conductivity of gas, and  $Q_S$  is the thermal emission at liquid solidification.

In Figure 31, dependences of the average temperature and radius of a jet on time for all cooling stages are presented. At calculation, the heat leakage has been taken as minimally as possible. In the figure, it is visible that within  $(15-25) \times 10^{-6}$  s (depending on diameter) after an output from the nozzle, the jet freezes completely. At characteristic liquid velocity of 10-20 m/s, the section of the jet that is on an output nozzle will depart on a distance of 0.15-0.5 mm. If the liquid velocity will be less, the front of solidification can come inside of the channel. The resistance to the movement of a jet, as a result, will increase, that else more will reduce the velocity. If the heat exchange with a wall of the channel will be insufficient, the process can end with full freezing of the channel. It is visible also that at full freezing of a liquid in a jet its diameter decreases by about 15%.

### 6.2. Experiments with a Jet. Choice of a Working Mode

Experiments on studying the behavior of a jet were realized at test of nozzle unit variants and for a choice a thermophysical mode at which the stable existence and the greatest stability of a jet are provided. Experiences were carried out for three inert gases – Ar, Kr, and Xe. As a result of experiments, the operating conditions of the nozzle unit for different gases have been certain. They are summarized in Table 9.

Table 9. Operating conditions of nozzle unit for different gases

Parameter	Argon	Kr	Xe
Temperature (K) Pressure (MPa)	85–95 0.1–0. 8	120–125 0.2–0.4	165–170 0.2–0.4



Fig. 32. A view of a condensed noble gases jet under vertical arrangement. (1) Ar, (2) Kr, (3) Xe.

Target Technology in RFNC-VNIIEF (Sarov)

#### 6.2.1. Experiments with a Vertical Arrangement of a Jet

The view of a jet of liquefied gases is shown on Figure 32. In these experiments, the nozzle variant with the long channel was used. The diameter of the jet is 20  $\mu$ m, the length of the capillary is 1 mm. During the work, the outflow of the jet was unstable in a direction; it chaotically changed the angle up to  $\pm 15^{\circ}$ . Perfection of nozzle design has allowed improving the stability of the jet: the deviation of the jet direction on a distance of 1 mm did not exceed its diameter. On a distance of 0.5 mm, the deviation is not appreciable. On a distance of 4 mm from the nozzle, the deviation of the jet can reach  $\pm 1$  mm. These deviations have chaotic character. On some frame, it is visible that a jet is bent. It is said that the jet is solid.

# 6.2.2. Experiments with a Horizontal Arrangement of a Jet

In Figure 33, photos of a jet of liquid Kr and Xe are presented. In this case, it used a shorter nozzle (diameter of





**Fig. 33.** A looks of jet of 20 mcm diameter under horizontal arrangement. (**a**) Kr (temperature 130–140 K, pressure 0.4–0.5 MPa), (**b**) Xe (temperature 170–180 K, pressure 0.4–0.5 MPa).



**Fig. 34.** Schematic image of temperature fluctuations of nozzle unit and states of jet. The shaded area is transitive when there is a change of aggregative state of jet substance.

 $10 \ \mu m$ , length of  $30 \ \mu m$ ) than in case of vertical arrangement. The velocity of the liquid equals about  $10 \ m/s$ . The jet is straight but changes the direction within the angle limits of  $15^{\circ}$ . These changes in the direction are connected mainly with the solid microparticles getting in the nozzle.

Research on the character of liquid jet outflow depending on temperature and pressure was carried out at cyclic change





Fig. 35. Photos of a Xe jet in liquid (a) and solid (b) state.



Fig. 36. Photo of the liquid Xe outflow through nozzle with three apertures.

of temperature. The character of the outflow varied as shown in Figure 34. Thus, pressure was supported to be constant. In experiments, the nozzle with a capillary of a diameter of 15  $\mu$ m and a length of 30  $\mu$ m was used, the working gas was Xe. The temperature slowly changed, the average value was nearly 181 K with amplitude of modification of  $\pm 2$  K. At pressure of 0.3 MPa and temperature above 181 K, the jet was liquid and moved with high velocity. At temperature below 181 K, the jet solidified and moved with small speed. Both cases are shown in Figure 35.

At increase of pressure, the time of finding the jet in solid state decreases and at a pressure larger than 0.8 MPa, the jet remained liquid always. Thus, the direction of a jet quickly and chaotically changes within the limits of a cone with angle nearby  $10^{\circ}$ .

Studying the work of the nozzle containing three capillaries was performed. The capillaries had a diameter of 10  $\mu$ m and were located on one straight line at a distance of 20  $\mu$ m from each other. The image of the liquid Xe outflow from such a nozzle is shown on Figure 36. It is visible that the jets deviate from each other. A parallel arrangement of all three jets was not successful.

Research of jet behavior at various variants of nozzle unit with the condenser has allowed to choose an optimal operating mode and to create a design with short nozzle. By means of such device, it was possible to generate a liquid Xe horizontal jet, which was stable during tens of minutes.

## 6.2.3. Experiments on Influence of the Laser Radiation on a Jet

Experiments on influence of the laser radiation on a vertical jet have shown that during an irradiation of a jet by powerful laser impulse on a distance of 1-1.5 mm from a capillary, the failure of the jet is observed. The jet was solid and had a small velocity. Therefore the shock wave arising at formation of the laser plasma had a strong influence. The jet is restored within 1-2 s. This process is shown in Figure 37.

At research of the influence of the pulse-periodic laser radiation on a horizontal jet, such a long failure was not observed. The jet was liquid and had greater velocity. The video in Figure 38 shows the outflow of three jets on which laser radiation influences from top. The frequency of shooting was 30 frames per second with time of exposition nearby 0.001 s, the frequency of laser pulse repetition is 25 Hz. The energy of radiation in an impulse was 0.05 J, the pulse duration was 10 ns.

From this frame, it is visible that the laser pulse breaks off the jet too or, at least, changes the direction (the jet becomes curved). Besides, it is visible that the plasma is displaced downward. However, the time of the jet restoration is much



Fig. 38. Picture of action of the laser radiation on three liquid jets.



Fig. 37. Video of jet failure under action of single laser pulse. (1) jet before pulse, (2) impact of a laser pulse on jet, (3) failure of jet, (4) beginning of jet restoration, (5) restored jet.

less than for the previous case. It makes value much less than 0.01 s. This value will limit the frequency of laser pulses repetition at use of jet as a target. To increase the frequency of laser influence it is required to increase the jet velocity.

#### 6.3. The Conclusion for Condensed Gases Jet Generator for Nanolithography

As a result of experiments, the efficiency of systems of cooling, evacuation and maintenance of working temperature, and pressure has been shown. The optimum design and an operating mode have allowed receiving a jet of liquid Xe that was stable during several tens of minutes. The study of the influence of laser pulses has shown that the time of a jet restoration after the impact of the laser pulse is much less than 0.01 s.

#### ACKNOWLEDGMENT

This work was financially supported by project ISTC #991.

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